

Photon Counting Module for Laser Time Transfer Space Mission

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GOALS

- Fast photon counting detectors for the
Laser Time Transfer space mission, China
- BACKGROUND
the K14 SPAD detectors have been launched onboard
MARS 96 (Russia) and NASA Mars Polar Lander (USA)
space missions
- REQUIREMENTS
 - low mass, power, bias voltage
 - high radiation in - sensitivity (> 5 years in space)
 - high temperature range
 - extreme optical damage threshold
(full Solar flux, no shutter)



„LTT Module in Space”, China, 2007-2008

- GOALs
- to synchronize the rubidium clocks in space, hydrogen masers in a future.
- Laser Time Transfer (LTT) between space and ground
- employing the existing China Satellite Laser Ranging network consisting of 5 fixed and 2 mobile systems
- required ~ 100 ps timing accuracy
- expected accuracy improvement
 >> 10x over RF techniques



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Detector Requirements - version LTT China



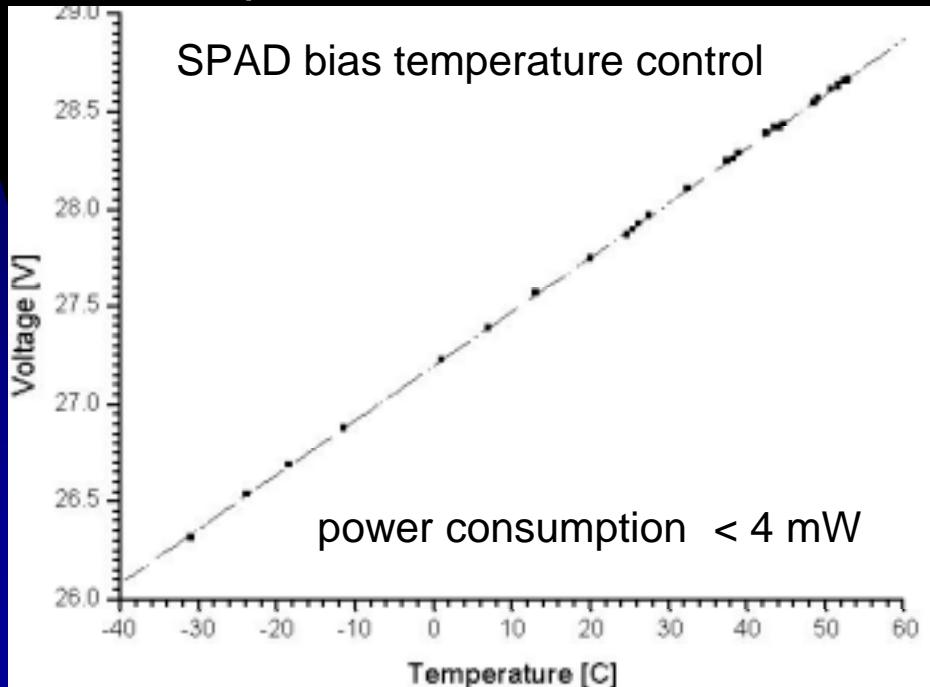
- single photon timing K14 SPAD chips
two channels
 - aperture 25 µm each
 - timing resolution < 100 psec
 - power, mass < 2 W , 100 grams
 - operating temperature -30 ...+60°C
 - lifetime in space > 5 years
 - high opt. damage threshold direct exposure to the Sun (!!)
in a focal plane of 2 mm aperture collecting optics
no Sun safety shutter will be installed
 - design & construction 3 months (!)

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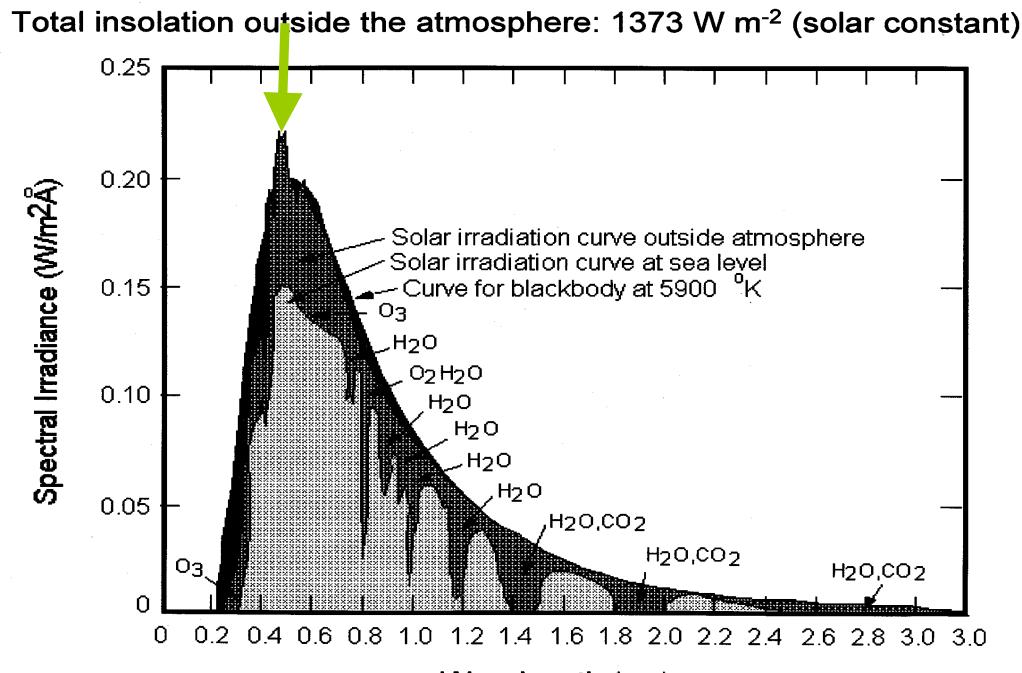
SPAD Bias Temperature Control

- SPAD break down voltage 29 Volts
- bias accuracy required 100 mV
- temperature range requested -30 ... + 60° C
no temperature control or cooling
- SPAD break voltage temperature drift - 30 mV / K
- => temperature controlled bias circuit



Optical Damage Threshold

Solar Spectrum



Source: The University of Texas at Austin, Center for Space Research

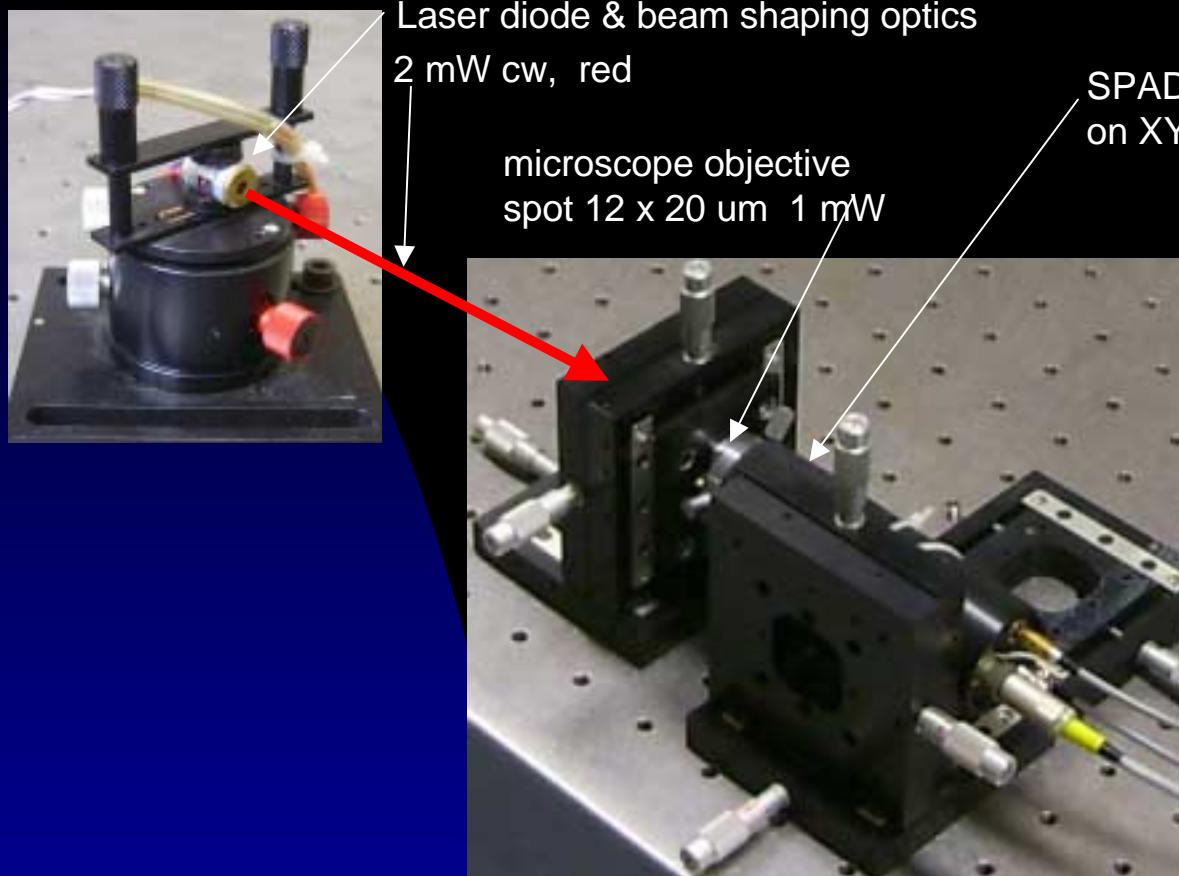
- Irradiance $0.2 \text{ W/m}^2/0.1 \text{ nm}$
@ 532nm wavelength
- receiver
 - aperture 2 mm
 - f / d ~ 1.0
 - field of view $\sim 0.5^\circ$
 - entire Solar disc
- bandwidth 100 nm
- blocking glass filter
- $\Rightarrow 1 \text{ mW max. on SPAD}$

Surprisingly, the total flux on the detector aperture is not exceeding 1 mW /100 nm for any aperture (!), due to the field of view limitation.

Larger telescope is not capable to focus all the incoming Sun light onto small SPAD aperture.

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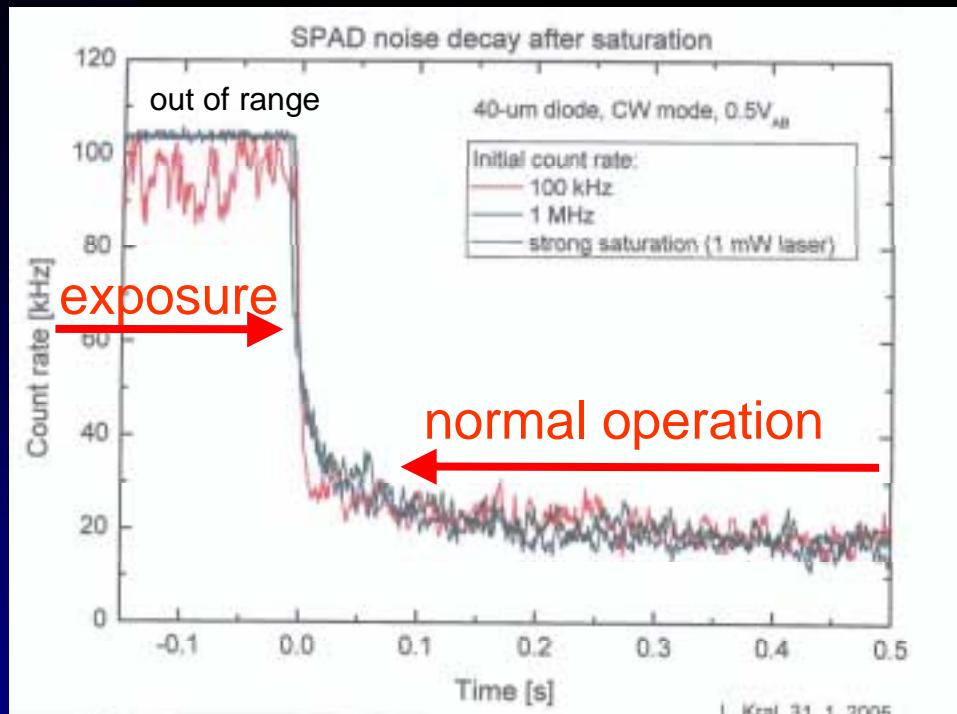
Optical Damage Tests



- exposure tests :
- no bias 3 x 8 hr
- biased 3 x 8 hr

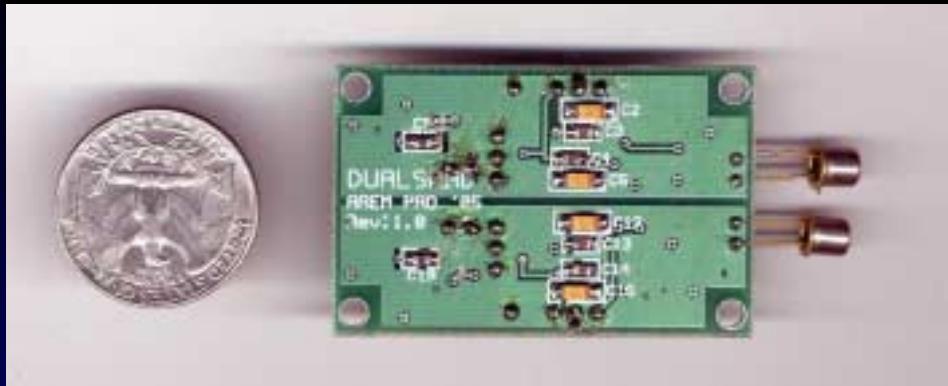
- NO detectable detector degradation after all optical irradiation tests
- Any size telescope with SPAD detector may be pointed toward the Sun without the damage (< 100 nm bandwidth)

Optical Saturation Recovery



- Detector operation recovery after strong optical signal exp.
- detector illumination
 - ambient light 100 kHz
 - attenuated laser 1 MHz
 - out of range when illuminated
 - full laser 1 mW NA
 - out of range when illuminated
- instrument time constant ~ 0.02 s
- Detector recovery time after saturation is well below 100 ms
- within this time, the dark count rate drops to 1.1 times the standard value

Photon Counting Module for Space Mission LTT



Technology demonstrator

Prague, March 2005



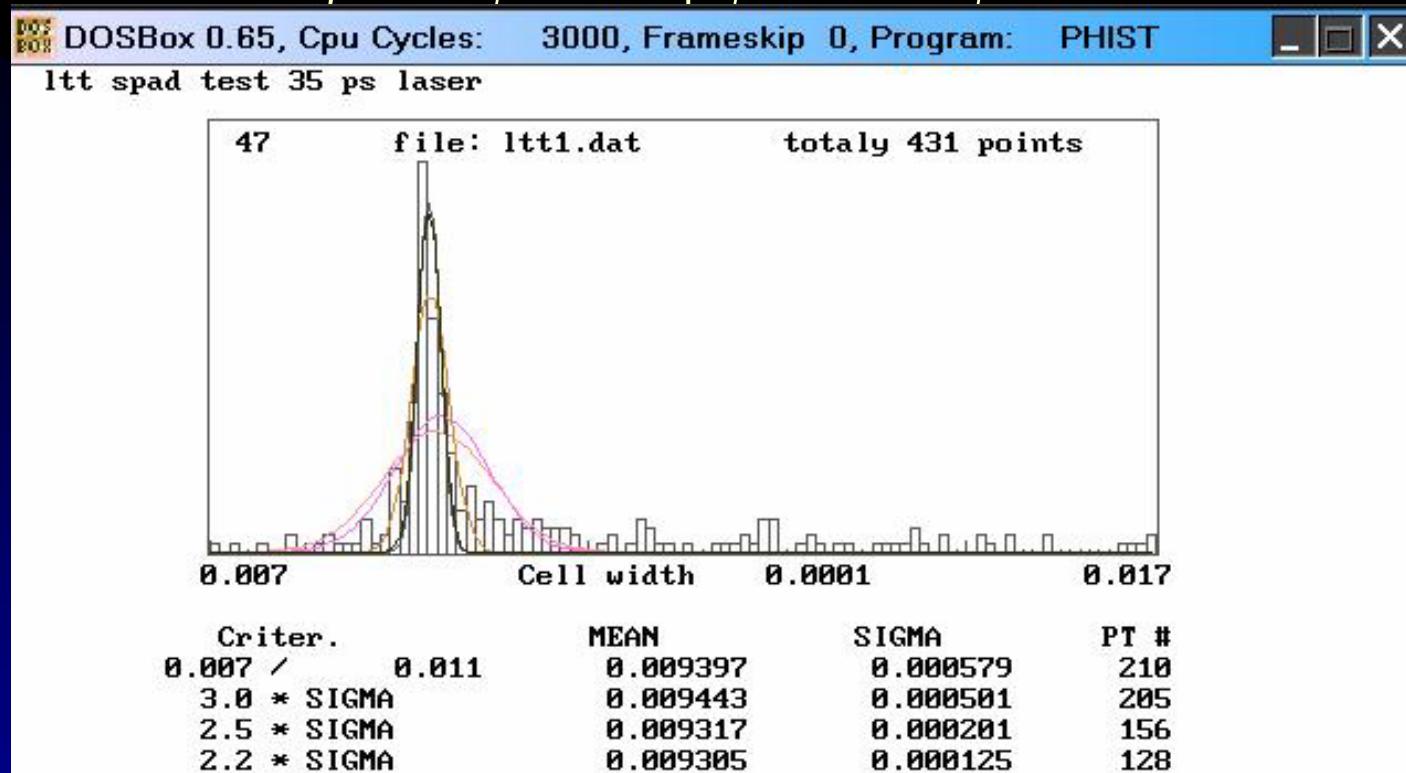
Detector package sample for pre-flight tests

Shanghai, China, July 2006

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SPAD Timing Resolution Tests, Shanghai July 2006

Shanghai SLR, laser 35 ps, HP counter, Detector # 1



- Jitter detector # 1 125 psec
- detector # 2 120 psec
- Detection delay difference 440 +/- 20 psec

Dual Single Photon Counting Module Detector Technology Demonstrator - Specifications

- | | |
|----------------------|--|
| ■ configuration | dual photon counting detector
based on Silicon K14 SPAD |
| ■ quenching | active |
| ■ active area | circular 25 um diameter |
| ■ quantum efficiency | ~ 10 % @ 532 nm |
| ■ timing resolution | 75 psec |
| ■ dark count rate | < 8 kHz @ +20°C |
| ■ operating temp. | -30 ... +60°C
no cooling, no stabilisation |
| ■ power consumption | < 400 mW |
| ■ mass | 4 grams |
| ■ optical damage th. | full Solar flux 100 nm BW, > 8 hr |
| ■ lifetime in space | > 10 years |



CONCLUSION

Photon Counting Module for Space Mission LTT

- the Technology Demonstrators have been completed
Prague, March 2005



- the Flight Unit detector version has been completed
Shanghai, July 2006
- Solar flux resistant using moderate wavelength filtering
- radiation resistant, 100 kRads without parameter change
=> lifetime in space > 10 years
- pre-flight tests, Shanghai, Beijin , fall 2006

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